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MEMORANISM FOR: Acting Chief, DeD-DD/P

SUBJECT:

Deliavilland Carebon DCH-4 (Army CA-1) Preliminary Operational Study

PEPERSONS:

- (A) U.S. Ammy Report, "User Tests of CA-1 Treasport Aircraft"
- (B) FAA Technical Report, "Project Hamminghird," Dated April 1961
- 1. This mesorender contains a recommendation for the approval of the Acting Chief, DPD-DD/P. Said recommendation is contained in paragraph 4.
- 2. The following information pertains to the Delinvilland "Carabou" DCH-4 (U.S. Army CA-1) and is based upon a study of above referenced material and field operating requirements of the control of the
 - a. American Gross Weight: 26,000 lbs. limit on first ten aircraft. This will be increased to 28,500 pounds beginning with aircraft No. 11 and subsequent.
 - b. Empty Weight of Test Aircreft: 18,230 lbs.
 - c. Capabilities:

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- (1) Carry 32 passengers or 28 combet-
- (2) Carry 3,000 lbs. of cargo 870 HM with reserve.
 - (3) Marchael renge 1,170 凝.
- (4) Carry 3 tons abort haul. Carry 7,500 pounds short haul (alreraft No. 11 and subsequent).
- (5) Aerial deliver heavy equipment and eight troops from rear exit (resp).
- (6) A-rial deliver one (1) stick of 24 troops from either side door, or 35 troops alternately through each side door (one side door each side of sircraft). Simultaneous delivery of two (2) sticks of troops is not feasible by use of any combination of doors due lack of separation. Simultaneous use of remp and side door exit is not feasible.
- (7) Aerial deliver one (1) stick of 12 troops in 22 seconds (side door) or one (1) stick of 24 troops in 35 seconds using alternately both side doors.
- (G) Adaptable to "pendulum" serial delivery system. Army used side buffer boards. Rear exit (tell) is 73.5 inches (minimum) by 74 inches.
 - (9) Suitable for IFR and night operations.

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(10) Rapid "turn-around time" due to truck bed height floor and year loading rasp.

d. Limitations or Other Mortaonings:

- (1) Cannot drop more than one (1) stick of troops at a time.
- (2) Cannot load tail gate (resp) when extended, but tail gate can be used for cargo tie does by closing first than shifting load to tail gate.
- (3) Not truly "world mobile" as is except with sugmented fuel. Maximum range is presently 1,170 MM. He record of flights with drop or fucelage type fuel tanks; however, fuselage tank installation would be a simple modification and is not considered a shortening except for the delay in minor engineering and installation.
- (b) No information available on characteristics of aircraft during crash or ditching.

 High wing and high engine nacelles are a disadvantage with respect to personnel survival in each case.
- (5) User tests did not include aircraft performance. Unable to verify Delievilland claims.
- (6) The Army recommends a minimum error of three. Deflavilland says one pilot can operate the aircraft.

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o. Other Comments:

- Army, state they are very satisfied with the aircraft. Although some shortcomings appeared, as is always the case with a new aircraft, the Carabon second to have fewer. It was not stated what these "shortcomings" were; however, those discrepancies listed in the original Army "User" test report were corrected, retested, and fixes were found suitable. No attempt was made to resolve the single stick troop delivery limitation, probably due to past experience with C-123, C-119, and C-150 similar problems which resulted in many studies, tests, and Pube Goldberg resolutions that did not increase capabilities from two to three sticks as desired.
- (2) PAA classed the Carebou as a short field aircraft (PA) rather than a true COUL type in its technical report Project Busning-bird, dated April 1961. Delievilland performance material quoted by the report agrees with other material thus far available and lists the following performance (standard day):
 - (a) Field Length: 1,020 feet
 - (56 knots) Lending Speed: 65 m.p.h.
 - (c) <u>Service Ceiling</u>: *27,500 ft.

^{*} At 26,000 lbs. Gross (At 28,500 lbs., would be 725 feet take-off roll; service ceiling, 26,000 feet.)

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- (4) Teke-off Run: (At maximum
- (e) Landing Run: (At maximum Gross weight, where wind) 535 feet (Connect: Exrimen gross at valid time of data was 26,000 pounds.)
- (3) Delievilland is to test a model with propeller reversing and a turboprop model of the Carabou for the Army in the near future. Army has no follow-on procurement plans to date for such configurations, and we would probably not be interested in turboprop equipment for

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- (4) Engines are Pratt & Whitney R-2000-4, came as C-54 aircraft.
- (5) Cost of the aircraft is approximately \$750,000 each with spares.
- 3. Conclusions: The Carebou, from a preliminary operational evaluation, appears to be suitable as a supplement to present single engine delio alreraft. It will not operate into all Helio SFOL airstrips in but would operate into many of these. The aircraft will do the job of a C-47 better and can land many places that a C-47 or C-46 should not utilize. Acrial delivery capabilities are better than a C-47, troop and cargo-wise, better than a C-46 cargo-wise, but about equal to the C-46 for a paratroop operation where simultaneous supply drop is not a requirement. It will do the job of six or more helio L-86s in less total time, with twin-engine reliability. It will carry about one-third the payload of a C-123 aircraft, but will land where a C-123 could not land because of field length.

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4. Googsandations: The Carebon sirerest characteristics are seversile careful to instant a further physical evaluation of operational expeditorations. It is, therefore, recommended that one or now persons of this branch visit an every installation for familiarication and decpretention to evaluate the alternat in light of field exercional considerature. This been selected to represent this Brench.

5. This recreasion is submitted as a mailining operational evalmation only. Additional comments will be salmitted as some information a accuallated.

> 25X1A Codes, Air Support Brench, 1970-1279

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